

# CELESTIAL MECHANICS STUDY GUIDE

## CELESTIAL MECHANICS STUDY GUIDE: UNLOCKING THE SECRETS OF ORBITAL MOTION

CELESTIAL MECHANICS STUDY GUIDE: EMBARKING ON THE STUDY OF CELESTIAL MECHANICS CAN BE A PROFOUNDLY REWARDING JOURNEY, DELVING INTO THE INTRICATE DANCE OF CELESTIAL BODIES GOVERNED BY GRAVITY. THIS COMPREHENSIVE GUIDE IS DESIGNED TO ILLUMINATE THE FUNDAMENTAL PRINCIPLES AND COMPLEX THEORIES THAT UNDERPIN OUR UNDERSTANDING OF ORBITAL DYNAMICS, PLANETARY MOTION, AND SPACEFLIGHT. WE WILL METICULOUSLY EXPLORE THE HISTORICAL EVOLUTION OF THIS FASCINATING FIELD, FROM KEPLER'S EMPIRICAL LAWS TO NEWTON'S UNIVERSAL LAW OF GRAVITATION, AND VENTURE INTO THE MODERN ERA WITH DISCUSSIONS ON PERTURBATION THEORY AND ORBITAL MANEUVERS. WHETHER YOU ARE A STUDENT GRAPPLING WITH COURSEWORK, A BUDDING ASTROPHYSICIST, OR SIMPLY AN ENTHUSIAST OF THE COSMOS, THIS RESOURCE AIMS TO PROVIDE CLARITY AND DEPTH. OUR EXPLORATION WILL COVER ESSENTIAL CONCEPTS, MATHEMATICAL FRAMEWORKS, AND PRACTICAL APPLICATIONS, ENSURING A THOROUGH GRASP OF CELESTIAL MECHANICS.

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## HISTORICAL FOUNDATIONS OF ORBITAL THEORY

THE QUEST TO UNDERSTAND THE MOVEMENT OF STARS AND PLANETS HAS CAPTIVATED HUMAN MINDS FOR MILLENNIA. EARLY ASTRONOMICAL OBSERVATIONS, DRIVEN BY THE NEED FOR CALENDARS AND NAVIGATION, LAID THE GROUNDWORK FOR WHAT WOULD BECOME CELESTIAL MECHANICS. ANCIENT CIVILIZATIONS METICULOUSLY TRACKED CELESTIAL OBJECT MOVEMENTS, DEVELOPING GEOCENTRIC MODELS THAT, WHILE ULTIMATELY INCORRECT, REPRESENTED SIGNIFICANT INTELLECTUAL ACHIEVEMENTS. THESE EARLY MODELS, SUCH AS PTOLEMY'S SYSTEM, ATTEMPTED TO EXPLAIN THE OBSERVED RETROGRADE MOTION OF PLANETS THROUGH COMPLEX EPICYCLES AND DEFERENTS. DESPITE THEIR LIMITATIONS, THESE EFFORTS ESTABLISHED A TRADITION OF RIGOROUS OBSERVATION AND MATHEMATICAL MODELING IN ASTRONOMY.

A PIVOTAL SHIFT OCCURRED WITH THE RENAISSANCE AND THE HELIOCENTRIC REVOLUTION. NICOLAUS COPERNICUS BRAVELY PROPOSED A SUN-CENTERED SOLAR SYSTEM, OFFERING A SIMPLER EXPLANATION FOR PLANETARY MOVEMENTS. HIS MODEL, HOWEVER, STILL RELIED ON PERFECT CIRCLES, NECESSITATING ADJUSTMENTS TO MATCH OBSERVATIONAL DATA. IT WAS THE RELENTLESS DEDICATION OF TYCHO BRAHE, WITH HIS UNPARALLELED PRECISION IN ASTRONOMICAL MEASUREMENTS, AND THE SUBSEQUENT WORK OF JOHANNES KEPLER THAT TRULY TRANSFORMED OUR UNDERSTANDING. KEPLER, BUILDING UPON BRAHE'S DATA, WAS ABLE TO DISTILL THE COMPLEX PATTERNS OF PLANETARY MOTION INTO ELEGANT, EMPIRICAL LAWS, FUNDAMENTALLY CHANGING THE TRAJECTORY OF CELESTIAL MECHANICS.

## NEWTON'S LAW OF UNIVERSAL GRAVITATION

THE MONUMENTAL BREAKTHROUGH IN CELESTIAL MECHANICS ARRIVED WITH SIR ISAAC NEWTON. IN HIS SEMINAL WORK, PRINCIPIA MATHEMATICA, NEWTON PRESENTED HIS LAW OF UNIVERSAL GRAVITATION, A UNIFYING THEORY THAT EXPLAINED BOTH TERRESTRIAL AND CELESTIAL MOTION. THIS LAW POSITS THAT EVERY PARTICLE OF MATTER IN THE UNIVERSE ATTRACTS EVERY OTHER PARTICLE WITH A FORCE THAT IS DIRECTLY PROPORTIONAL TO THE PRODUCT OF THEIR MASSES AND INVERSELY PROPORTIONAL TO THE SQUARE OF THE DISTANCE BETWEEN THEIR CENTERS. MATHEMATICALLY, THIS IS EXPRESSED AS  $F = G \frac{m_1 m_2}{r^2}$

$\frac{m_1 m_2}{r^2}$ ), where  $(F)$  is the gravitational force,  $(G)$  is the gravitational constant,  $(m_1)$  and  $(m_2)$  are the masses of the two objects, and  $(r)$  is the distance between their centers.

Newton's law provided a powerful framework for predicting the motion of celestial bodies with unprecedented accuracy. It explained why objects fall to Earth and why the Moon orbits our planet, unifying these seemingly disparate phenomena under a single, elegant principle. This universal law not only validated Kepler's empirical observations but also provided the underlying physical mechanism for them. The ability to mathematically describe and predict the orbits of planets, comets, and moons was a profound testament to the predictive power of Newton's physics, laying the foundation for modern astrophysics and space exploration.

## KEPLER'S LAWS OF PLANETARY MOTION

Johannes Kepler, through meticulous analysis of Tycho Brahe's extensive observational data, formulated three fundamental laws that describe the motion of planets around the Sun. These laws, derived empirically, were groundbreaking and provided the first accurate mathematical description of orbital paths. They marked a departure from the ancient belief in perfect circular motion, ushering in an era of elliptical orbits.

- **KEPLER'S FIRST LAW (LAW OF ELLIPSES):** The orbit of every planet is an ellipse with the Sun at one of the two foci. This law was revolutionary, replacing the notion of perfect circles with the more accurate description of ellipses for planetary paths.
- **KEPLER'S SECOND LAW (LAW OF EQUAL AREAS):** A line joining a planet and the Sun sweeps out equal areas during equal intervals of time. This implies that planets move faster when they are closer to the Sun (perihelion) and slower when they are farther away (aphelion).
- **KEPLER'S THIRD LAW (LAW OF HARMONIES):** The square of the orbital period of a planet is directly proportional to the cube of the semi-major axis of its orbit. This law connects the orbital period of a planet to its average distance from the Sun, revealing a harmonious relationship in the solar system.

These laws, though empirical, were a profound achievement and served as crucial empirical evidence that Newton later used to derive his law of universal gravitation. Understanding Kepler's laws is fundamental to grasping the basics of orbital mechanics and the predictable, yet elegant, movements of celestial bodies within a gravitational system.

## THE TWO-BODY PROBLEM

The two-body problem is a cornerstone of celestial mechanics, simplifying complex gravitational interactions by considering only two point masses interacting with each other. This idealized scenario assumes that these two bodies are isolated, with no other significant gravitational influences. In the context of celestial mechanics, the most common example is the Earth-Moon system or a satellite orbiting a planet, where the mass of the orbiting body is negligible compared to the central body.

Solving the two-body problem allows us to determine the exact trajectory of one body around another under their mutual gravitational attraction. The solution reveals that the possible orbits are conic sections: ellipses (bound orbits), parabolas (escape trajectories), or hyperbolas (unbound trajectories). For most celestial bodies in stable orbits, such as planets around a star, the orbits are elliptical. The mathematical framework for solving the two-body problem involves applying Newton's laws of motion and gravitation, typically resulting in differential equations that can be integrated to yield the position and velocity of the bodies as a function of time.

# ELLIPTICAL ORBITS AND ORBITAL ELEMENTS

ELLIPTICAL ORBITS ARE THE MOST COMMON TYPE OF BOUND TRAJECTORIES OBSERVED IN CELESTIAL MECHANICS, DESCRIBING THE PATHS OF PLANETS AROUND STARS, MOONS AROUND PLANETS, AND SATELLITES AROUND EARTH. EACH ELLIPTICAL ORBIT CAN BE UNIQUELY DEFINED BY A SET OF SIX ORBITAL ELEMENTS, WHICH PROVIDE A COMPLETE DESCRIPTION OF THE OBJECT'S PATH AND POSITION IN SPACE AT ANY GIVEN TIME. THESE ELEMENTS ARE DERIVED FROM THE SOLUTION TO THE TWO-BODY PROBLEM.

THE SIX CLASSICAL ORBITAL ELEMENTS ARE:

- **SEMI-MAJOR AXIS (A):** THIS DEFINES THE SIZE OF THE ELLIPSE. FOR AN ELLIPSE, IT'S HALF THE LONGEST DIAMETER. IN CELESTIAL MECHANICS, IT'S OFTEN RELATED TO THE ENERGY OF THE ORBIT.
- **ECCENTRICITY (E):** THIS DEFINES THE SHAPE OF THE ELLIPSE, RANGING FROM 0 FOR A PERFECT CIRCLE TO LESS THAN 1 FOR AN ELLIPSE. A HIGHER ECCENTRICITY MEANS A MORE ELONGATED ORBIT.
- **INCLINATION (I):** THIS IS THE ANGLE BETWEEN THE ORBITAL PLANE AND A REFERENCE PLANE (E.G., THE EARTH'S EQUATORIAL PLANE OR THE ECLIPTIC PLANE).
- **LONGITUDE OF THE ASCENDING NODE ( $\Omega$ ):** THIS SPECIFIES THE ORIENTATION OF THE ORBITAL PLANE IN SPACE, MEASURING THE ANGLE FROM A REFERENCE DIRECTION (E.G., THE VERNAL EQUINOX) TO THE POINT WHERE THE ORBIT CROSSES THE REFERENCE PLANE FROM SOUTH TO NORTH.
- **ARGUMENT OF PERIAPSIS ( $\omega$ ):** THIS DEFINES THE ORIENTATION OF THE ELLIPSE WITHIN ITS ORBITAL PLANE, MEASURING THE ANGLE FROM THE ASCENDING NODE TO THE PERIAPSIS (THE POINT OF CLOSEST APPROACH TO THE CENTRAL BODY).
- **TRUE ANOMALY (N) OR MEAN ANOMALY (M) OR TRUE LONGITUDE ( $\lambda$ ):** THIS SPECIFIES THE POSITION OF THE BODY ALONG ITS ORBIT AT A PARTICULAR TIME. TRUE ANOMALY IS THE ANGLE MEASURED FROM PERIAPSIS TO THE BODY'S CURRENT POSITION, WHILE MEAN ANOMALY IS A TIME-BASED ANGLE.

UNDERSTANDING THESE ORBITAL ELEMENTS IS CRUCIAL FOR PREDICTING FUTURE POSITIONS OF CELESTIAL BODIES, PLANNING SPACE MISSIONS, AND ANALYZING OBSERVATIONAL DATA. THEY PROVIDE A STANDARDIZED WAY TO COMMUNICATE AND ANALYZE ORBITAL INFORMATION.

## PERTURBATION THEORY IN CELESTIAL MECHANICS

WHILE THE TWO-BODY PROBLEM PROVIDES AN EXCELLENT APPROXIMATION FOR ORBITAL MOTION, REAL-WORLD SCENARIOS ARE FAR MORE COMPLEX DUE TO THE GRAVITATIONAL INFLUENCE OF OTHER CELESTIAL BODIES. PERTURBATION THEORY IS THE BRANCH OF CELESTIAL MECHANICS THAT DEALS WITH THESE DEVIATIONS FROM IDEAL TWO-BODY ORBITS. THESE DEVIATIONS, OR PERTURBATIONS, CAUSE THE ORBITS TO CHANGE OVER TIME, DEVIATING FROM PERFECT CONIC SECTIONS.

PERTURBATIONS CAN ARISE FROM SEVERAL SOURCES. THE MOST SIGNIFICANT IS THE GRAVITATIONAL PULL OF OTHER PLANETS, STARS, OR MOONS IN THE SYSTEM, WHICH ACTS TO ALTER THE PRIMARY ORBIT. OTHER FACTORS INCLUDE NON-SPHERICAL SHAPES OF CELESTIAL BODIES, ATMOSPHERIC DRAG FOR SATELLITES CLOSE TO EARTH, AND RELATIVISTIC EFFECTS FOR BODIES IN STRONG GRAVITATIONAL FIELDS. PERTURBATION THEORY EMPLOYS VARIOUS ANALYTICAL AND NUMERICAL TECHNIQUES TO CALCULATE THESE SMALL DEVIATIONS AND THEIR CUMULATIVE EFFECTS ON THE ORBIT. THIS IS ESSENTIAL FOR LONG-TERM PREDICTIONS OF PLANETARY POSITIONS, THE STABILITY OF THE SOLAR SYSTEM, AND THE PRECISE NAVIGATION OF SPACECRAFT.

## N-BODY PROBLEM AND ITS COMPLEXITIES

THE N-BODY PROBLEM IS THE GENERALIZATION OF THE TWO-BODY PROBLEM TO THREE OR MORE BODIES. IN THIS SCENARIO, EACH BODY GRAVITATIONALLY INTERACTS WITH EVERY OTHER BODY IN THE SYSTEM SIMULTANEOUSLY. UNLIKE THE TWO-BODY PROBLEM, WHICH HAS AN EXACT ANALYTICAL SOLUTION, THE GENERAL N-BODY PROBLEM DOES NOT HAVE A CLOSED-FORM SOLUTION FOR  $N > 2$ . THIS MAKES IT SIGNIFICANTLY MORE CHALLENGING TO SOLVE AND PREDICT THE LONG-TERM BEHAVIOR OF SUCH SYSTEMS.

THE COMPLEXITY OF THE N-BODY PROBLEM ARISES FROM THE INTRICATE AND OFTEN CHAOTIC INTERACTIONS BETWEEN THE BODIES. EVEN SMALL INITIAL CHANGES IN THE POSITIONS OR VELOCITIES OF THE BODIES CAN LEAD TO DRAMATICALLY DIFFERENT LONG-TERM OUTCOMES, A HALLMARK OF CHAOTIC SYSTEMS. FOR INSTANCE, THE LONG-TERM STABILITY OF OUR OWN SOLAR SYSTEM, WITH ITS MULTIPLE PLANETS, IS A MANIFESTATION OF THE N-BODY PROBLEM. ASTRONOMERS AND PHYSICISTS RELY HEAVILY ON NUMERICAL SIMULATIONS AND APPROXIMATIONS TO STUDY THE N-BODY PROBLEM, USING POWERFUL COMPUTERS TO INTEGRATE THE EQUATIONS OF MOTION STEP-BY-STEP OVER EXTENDED PERIODS. UNDERSTANDING THESE COMPLEX INTERACTIONS IS VITAL FOR FIELDS LIKE GALACTIC DYNAMICS AND THE STUDY OF PLANETARY SYSTEM FORMATION.

## ORBITAL MANEUVERS AND SPACEFLIGHT APPLICATIONS

THE PRINCIPLES OF CELESTIAL MECHANICS ARE NOT JUST THEORETICAL; THEY ARE THE VERY FOUNDATION OF SPACEFLIGHT. ORBITAL MANEUVERS ARE PRECISELY CALCULATED ADJUSTMENTS TO A SPACECRAFT'S TRAJECTORY, ENABLING IT TO CHANGE ITS ORBIT, RENDEZVOUS WITH OTHER OBJECTS, OR ACHIEVE SPECIFIC MISSION OBJECTIVES. THESE MANEUVERS ARE BASED ON THE APPLICATION OF NEWTON'S LAWS AND THE UNDERSTANDING OF ORBITAL MECHANICS DERIVED FROM THE TWO-BODY PROBLEM AND PERTURBATION THEORY.

KEY ORBITAL MANEUVERS INCLUDE:

- **HOHMANN TRANSFER ORBIT:** A FUEL-EFFICIENT MANEUVER USED TO TRANSFER A SPACECRAFT BETWEEN TWO CIRCULAR ORBITS IN THE SAME PLANE. IT INVOLVES FIRING ENGINES TO MOVE THE SPACECRAFT INTO AN ELLIPTICAL TRANSFER ORBIT THAT IS TANGENT TO BOTH THE INITIAL AND FINAL ORBITS.
- **RENDEZVOUS AND DOCKING:** THE COMPLEX PROCESS OF BRINGING TWO SPACECRAFT INTO CLOSE PROXIMITY, OFTEN REQUIRING MULTIPLE PRECISE BURNS TO MATCH VELOCITIES AND POSITIONS.
- **GRAVITY ASSIST (SLINGSHOT MANEUVER):** A TECHNIQUE WHERE A SPACECRAFT USES THE GRAVITATIONAL PULL OF A PLANET TO ALTER ITS SPEED AND TRAJECTORY, ALLOWING IT TO TRAVEL TO DISTANT PARTS OF THE SOLAR SYSTEM WITH LESS FUEL.
- **ORBIT RAISING AND LOWERING:** USING ENGINE BURNS TO INCREASE OR DECREASE THE ALTITUDE OF A SPACECRAFT'S ORBIT.

THE ABILITY TO PREDICT AND EXECUTE THESE MANEUVERS IS CRITICAL FOR SUCCESSFUL SPACE MISSIONS, FROM LAUNCHING SATELLITES TO EXPLORING OTHER PLANETS. CELESTIAL MECHANICS PROVIDES THE ESSENTIAL TOOLS AND UNDERSTANDING TO NAVIGATE THE VASTNESS OF SPACE.

## MODERN CELESTIAL MECHANICS AND FUTURE DIRECTIONS

MODERN CELESTIAL MECHANICS CONTINUES TO EVOLVE, DRIVEN BY ADVANCES IN COMPUTATIONAL POWER, OBSERVATIONAL TECHNIQUES, AND THEORETICAL UNDERSTANDING. WHILE CLASSICAL NEWTONIAN MECHANICS REMAINS FUNDAMENTAL, NEW AREAS OF RESEARCH ARE PUSHING THE BOUNDARIES OF OUR KNOWLEDGE. THIS INCLUDES THE STUDY OF CHAOTIC DYNAMICS IN N-BODY SYSTEMS, THE DEVELOPMENT OF MORE SOPHISTICATED PERTURBATION METHODS, AND THE APPLICATION OF GENERAL RELATIVITY TO CELESTIAL PHENOMENA, ESPECIALLY IN STRONG GRAVITATIONAL FIELDS LIKE THOSE AROUND BLACK HOLES AND NEUTRON STARS.

FUTURE DIRECTIONS ALSO INVOLVE LEVERAGING MACHINE LEARNING AND ARTIFICIAL INTELLIGENCE FOR ANALYZING VAST DATASETS FROM ASTRONOMICAL SURVEYS AND FOR OPTIMIZING SPACECRAFT TRAJECTORIES. THE ONGOING DISCOVERY OF EXOPLANETS HAS OPENED UP NEW FRONTIERS IN STUDYING THE ORBITAL MECHANICS OF EXTRASOLAR SYSTEMS, PRESENTING UNIQUE CHALLENGES AND OPPORTUNITIES FOR UNDERSTANDING PLANETARY FORMATION AND STABILITY ACROSS THE GALAXY. AS HUMANITY CONTINUES TO EXPLORE SPACE, THE PRINCIPLES OF CELESTIAL MECHANICS WILL REMAIN INDISPENSABLE, GUIDING OUR ENDEAVORS AND DEEPENING OUR COMPREHENSION OF THE UNIVERSE'S GRAND DESIGN.

## FAQ

### Q: WHAT ARE THE MOST IMPORTANT CONCEPTS TO FOCUS ON WHEN STUDYING CELESTIAL MECHANICS?

A: THE MOST IMPORTANT CONCEPTS INCLUDE NEWTON'S LAW OF UNIVERSAL GRAVITATION, KEPLER'S LAWS OF PLANETARY MOTION, THE SOLUTION TO THE TWO-BODY PROBLEM, AND THE DEFINITION AND SIGNIFICANCE OF ORBITAL ELEMENTS. UNDERSTANDING PERTURBATION THEORY AND THE CHALLENGES OF THE N-BODY PROBLEM IS ALSO CRUCIAL FOR A COMPREHENSIVE GRASP OF THE SUBJECT.

### Q: HOW DOES CELESTIAL MECHANICS APPLY TO REAL-WORLD SPACE MISSIONS?

A: CELESTIAL MECHANICS IS FUNDAMENTAL TO SPACE MISSIONS. IT ENABLES THE CALCULATION OF LAUNCH WINDOWS, THE DESIGN OF ORBITAL TRAJECTORIES, THE EXECUTION OF ORBITAL MANEUVERS FOR CHANGING ORBITS OR RENDEZVOUS, AND THE USE OF GRAVITY ASSISTS TO TRAVEL EFFICIENTLY THROUGHOUT THE SOLAR SYSTEM. WITHOUT CELESTIAL MECHANICS, SPACEFLIGHT AS WE KNOW IT WOULD NOT BE POSSIBLE.

### Q: WHAT IS THE DIFFERENCE BETWEEN THE TWO-BODY PROBLEM AND THE N-BODY PROBLEM?

A: THE TWO-BODY PROBLEM SIMPLIFIES INTERACTIONS TO ONLY TWO CELESTIAL BODIES, ALLOWING FOR AN EXACT ANALYTICAL SOLUTION THAT DESCRIBES CONIC SECTION ORBITS. THE N-BODY PROBLEM, HOWEVER, CONSIDERS THE GRAVITATIONAL INTERACTIONS OF THREE OR MORE BODIES, MAKING IT SIGNIFICANTLY MORE COMPLEX WITH NO GENERAL ANALYTICAL SOLUTION, OFTEN REQUIRING NUMERICAL SIMULATIONS.

### Q: HOW ARE ELLIPTICAL ORBITS DESCRIBED IN CELESTIAL MECHANICS?

A: ELLIPTICAL ORBITS ARE DESCRIBED BY SIX ORBITAL ELEMENTS: THE SEMI-MAJOR AXIS (SIZE), ECCENTRICITY (SHAPE), INCLINATION (TILT OF THE ORBITAL PLANE), LONGITUDE OF THE ASCENDING NODE (ORIENTATION OF THE PLANE), ARGUMENT OF PERIAPSIS (ORIENTATION WITHIN THE PLANE), AND A MEASURE OF POSITION ALONG THE ORBIT AT A GIVEN TIME, SUCH AS TRUE ANOMALY OR MEAN ANOMALY.

### Q: WHAT IS PERTURBATION THEORY IN CELESTIAL MECHANICS?

A: PERTURBATION THEORY IS A METHOD USED TO STUDY THE DEVIATIONS OF ACTUAL CELESTIAL ORBITS FROM IDEAL TWO-BODY ORBITS. THESE DEVIATIONS, OR PERTURBATIONS, ARE CAUSED BY THE GRAVITATIONAL INFLUENCES OF OTHER CELESTIAL BODIES OR BY NON-IDEAL FACTORS LIKE ATMOSPHERIC DRAG, AND THEY CAUSE ORBITS TO CHANGE OVER TIME.

### Q: WHY IS THE N-BODY PROBLEM SO DIFFICULT TO SOLVE?

A: THE N-BODY PROBLEM IS DIFFICULT TO SOLVE ANALYTICALLY BECAUSE THE GRAVITATIONAL FORCE EXERTED BY EACH BODY ON EVERY OTHER BODY CHANGES CONSTANTLY AND IN A COMPLEX, INTERCONNECTED WAY. THIS INTERCONNECTEDNESS LEADS

TO INTRICATE DYNAMICS, AND FOR MORE THAN TWO BODIES, THESE INTERACTIONS CANNOT BE NEATLY CAPTURED BY A SINGLE SET OF EQUATIONS THAT YIELD A GENERAL SOLUTION.

## **Q: WHAT ARE SOME OF THE HISTORICAL FIGURES CRUCIAL TO THE DEVELOPMENT OF CELESTIAL MECHANICS?**

A: KEY HISTORICAL FIGURES INCLUDE NICOLAUS COPERNICUS, WHO PROPOSED THE HELIOCENTRIC MODEL; JOHANNES KEPLER, WHO FORMULATED THE EMPIRICAL LAWS OF PLANETARY MOTION; AND SIR ISAAC NEWTON, WHO DEVELOPED THE LAW OF UNIVERSAL GRAVITATION, PROVIDING THE PHYSICAL BASIS FOR KEPLER'S LAWS.

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